

Reality Expansion Pack for X-Plane

Robin DR400-180

Checklists & References

BEFORE ENGINE START

 Internal and external inspection 	COMPLETED
2. Canopy	AS REQUIRED
3. Flaps	UP
4. Seatbelts	FASTEN
5. Avionics	OFF
6. Parking Brake	SET
7. Passenger Briefing	COMPLETED

ENGINE START

	CLEAR
2. Carb. Heat	COLD
3. Throttle	IDLE
4. Mixture	RICH
5. Master Battery	ON
6. Anti-collision Light	ON
7. Fuel Pump	ON
8. Fuel Pressure	CHECK
	LEFT
	PUMP THREE TIMES
11. Throttle	OPEN 1/2 INCH
12. Starter	ENGAGE
When engine starts:	
13. Magnetos	BOTH
	1000RPM
15. Oil Pressure	CHECK
16. Fuel Pump	OFF
17. Alternator	ON
18. Amperometer	CHECK
	OFF
	I FAN AS REQUIRED

BEFORE TAXI

1. Fuel Selector	LEFT	
2. Warning Lights	TEST	
3. Avionics	ON AND SET	
4. Flight Instruments	CHECK AND SET	
5. Flight Controls	TEST	
6. Taxi Briefing	COMPLETED	
7. Fuel Selector	RIGHT	
TAXI		
1. Brakes	TEST	
2. Gyro Instruments and Compass	CHECK	

ENGINE RUN-UP

1. Parking Brake	SET
2. Fuel Selector	FULLEST TANK
3. Oil Temperature	> 40°C
4. Mixture	RICH
5. Throttle	2000RPM
6. Magnetos	CHECK (175-50 RPM)
7 Carla Haat	CHECK
3. Mixture	CHECK THEN RICH
9. Engine Instr. and Amp	CHECK
10. Vacuum	CHECK
11. Throttle	IDLE THEN CHECK 600-650RPM
12. Throttle	1000RPM

BEFORE TAKE-OFF

1. Canopy	CLOSED AND LATCHED
2 Magnetes	BOTH
3. Mixture	RICH
4. Carb Heat	COLD
F. Fuel Dump	ON
6. Flaps	1 NOTCH
7. Trim	SET
8. Take-off Briefing	COMPLETED
9. Landing Light	ON
10. Transponder	

TAKE-OFF

1. Throttle	FULL FORWARD (MIN. 2200RPM)
2. Rotation at	100 KM/HR
3. Climb at Vx	120 KM/HR
At 500ft AGL:	
4. Accelerate to Vy	150/170 KM/HR
5. Flaps	UP
6. Fuel Pump	OFF
7. Landing Light	OFF
	CLIMB
1. Throttle	FULL FORWARD (Max 2700RPM)
2. Mixture	LEAN AS REQUIRED
3. Engine Instruments	CHECK
4 Gyro and Compass	CHECK

CRUISE

1. Throttle	AVOID RED ARC
2. Mixture	AS REQUIRED
3. Gyro and Compass	CHECK
4. Engine Instr. and Amp.	CHECK
5. Fuel Comsumption	CHECK
FUEL TANK SW	ITCH
1. Fuel Pump	ON
2. Fuel Selector	SWITCH
3. Fuel Pump	OFF
4. Fuel Pressure	MONITOR

DESCENT

	COMPLETED AS REQUIRED FULLEST TANK	
APPROACH		
1. FLAP (< 170 Km/hr)	1 NOTCH	
O. Mindama	RICH	
3. Carb Heat	НОТ	
4. Fuel Pump		
5. Landing Light	ON	
6. Speed	150 KM/HR	
FINAL		
1. Flaps	FULL	
2. Carb Heat		
	130 KM/HR + 1/2 GUST	

AFTER LANDING

1. Flaps	UP
2. Trim	
3. Carb Heat	
4. Transponder	
5. Fuel Pump	
6. Landing Light	OFF
PARKING	
1. Parking Brake	ON
2. Avionics	OFF
3. Throttle	1000RPM
4. Mixture	CUT-OFF
5. Magnetos	OFF
6. Eletrical Switches	
7. Master Battery	
8. Flaps	FULL

ENGINE FAILURE ON TAKE-OFF

1. Speed	AVOID STALL	
2. Fuel Selector	CLOSE	
	CLOSE	
	OFF	
5. Magnetos	OFF	
ENGINE FIRE IN FLIGHT		
1. Fuel Selector	CLOSE	
2. Throttle	FULL FORWARD	
3. Mixture	CUT-OFF	
4. Fuel Pump	OFF	
5. Cabin Heat	OFF	
	Vno - 260 KM/HR	
7. Best Glide	150 KM/HR	
8. Cockpit	PREPARE FOR FORCED LANDING	

ENGINE FAILURE IN FLIGHT

1. Best Glide (No Fla	p) 150 KM/HR
2. Fuel Selector	OPEN
3. Mixture	FULL RICH
4. Carb Heat	HOT
5. Throttle	OPEN
6. Fuel Pump	ON
7. Magnetos	ВОТН
8. Cockpit	PREPARE FOR FORCED LANDING

FORCED LANDING

1. Seatbelts	FASTEN
2. Fuel Selector	CLOSED
3. Mixture	CUT-OFF
4. Throttle	IDLE
5. Fuel Pump	OFF
6. Magnetos	OFF
On final:	
7. Flaps	FULL
8. Battery and Alternator	OFF
9. Canopy	UNLATCH

NOISE LIMITATION

In compliance with the decree of 19.02.1987, the maximum acceptable noise level for the DR400/180 aircraft, at a certified gross weight of (2425 lb) 1100 kg is 84.6 dB(A) (ICAO annex 16 chapter 10).

The noise level determined under the conditions of the decree, is 76.4 dB(A) at max. continuous power.

The DR400/180 aircraft has received noise limitation certificate nr N45.

AIRSPEED INSTALLATION CALIBRATION

VC = (VI + calibration) is substantially equal to VI

The above figures do not take into account the ASI own tolerance.

NOTE

All speeds in this manual are Indicated Air speeds unless otherwise specified.

STALL SPEEDS

Weight 1100 kg (2425 lb) engine idle	km/h (kt)						
Bank angle	0°	30°	60°				
Flaps up	105 (57)	113 (61)	148 (78)				
Flaps Take off position	99 (53)	106 (57)	140 (76)				
Flaps Landing position	95 (51)	102 (55)	134 (72)				

TAKE OFF PERFORMANCE

At gross weight 1100 kg (2425 lb) Without wind, flaps in "take off position" (1st notch), engine full power.

Take off speed......(54 kt) 100 km/h Over 15 m (50 ft) barrier speed.....(70 kt) 130 km/h

Pressure Altitude (ft)	Temperature °C (°F)	Weight 1100 Kg (2425 lb)					We 900 kg (ight 1984 lb)	
			e off ance		o clear ft) barrier	Take off distance		Run to clear 15m(50ft) barrier	
		m	(ft)	m	(ft)	m	(ft)	m	(ft)
0	- 5 (23)	215	(700)	445	(1450)	120	(395)	250	(820)
	Std = 15 (59)	250	(815)	515	(1690)	140	(460)	290	(955)
	35 (95)	290	(945)	600	(1955)	165	(535)	340	(1105)
2500	- 10 (14)	260	(860)	540	(1780)	150	(485)	310	(1005)
	Std = 10 (50)	305	(1005)	635	(2085)	175	(565)	360	(1175)
	30 (86)	355	(1165)	735	(2415)	200	(655)	415	(1360)
5000	- 15 (5)	330	(1075)	680	(2225)	185	(605)	385	(1255)
	Std = 5 (41)	385	(1260)	795	(2610)	215	(710)	450	(1475)
	25 (77)	445	(1465)	925	(3035)	250	(825)	520	(1710)
8000	- 21 (-6)	430	(1410)	890	(2925)	245	(795)	505	(1660)
	Std = -1 (30)	505	(1660)	1050	(3445)	285	(940)	590	(1945)
	19 (66)	590	(1935)	1225	(4010)	335	(1095)	695	(2265)

Head wind influence:

For 10 kt multiply by 0.85

For 20 kt multiply by 0.65 For 30 kt multiply by 0.55

Down wind influence:

Add 10% to distance per section of 2 kt

Dried grass runway:

Add 15%

CLIMB PERFORMANCE

1) Flaps, take off position:

At maximum weight of 1100 kg (2425 lb) in standard atmosphere

Maximum rate of climb after take off	(827 ft/mn) 4.2 m/s
reduct	ion of 0.24 m/s (47 ft/mn) per 1000 ft
Best rate of climb speed	(81 kt) 150 km/h
Best angle of climb speed	(70 kt) 130 km/h

2) Flaps up:

In standard atmosphere, Full throttle, mixture best power,

- At maximum weight of 1100 kg (2425 lb):

Maximum rate of climb after take off	(885 ft/mn) 4.5 m/s
reduct	tion of 0.24 m/s (47 ft/mn) per 1000 ft
Service ceiling	14720 ft
Best rate of climb speed after take of	ff(92 kt) 170 km/h)
	up to ceiling (86 kt) 160 km/h
Best angle of climb speed	(76 kt) 140 km/h

- At weight of 900 kg (1984 lb):

Maximum rate of climb after take	off (1200 ft/mn) 6.1 m/s
redu	uction of 0.26 m/s (51 ft/mn) per 1000 ft
Service ceiling	19720 ft

Temperature influence:

Each 10°C above standard, lowers the ceiling by 1000 ft and reduces rate of climb by 0.24 m/s (47 ft/mn).

Time, Consumption, Climb distance

At gross weight 1100 kg (2425 lb)
Without wind, in standard atmosphere,
Flaps retracted, full power: maximum rate of climb after take off.
Start and roll consumption included.

PRESSURE ALTITUDE ZP (ft)	TIME (min)	FUEL CONSUMPTION I (imp/us gal)	RAI (km)	NGE (Nm)
3000	4	4.5 (1/1.2)	9.3	5
5500	7.5	8 (1.8/2.1)	17.6	9.5
8500	16.5	15 (3.3/4)	38.8	21

Glide performance

Engine off, the aircraft glides 9,3 time its height (without wind) at 150 km/h (81 kt).

Altitude and temperature do not have a perceptible influence.

CRUISE PERFORMANCE

At gross weight 1100 kg (2425 lb), in standard atmosphere. Optimum mixture setting, usable fuel (41.58 imp/49.1 us gal) 189 l . Without reserve fuel, without wind.

Consumption and climbing time compensated with descent.

ALTI- TUDE	PO	WER	FUEL CONSUMPTION		TRUE AIR SPEED		ENDU- RANCE	RANGE		
Zp(ft)	%	rpm	l/h	gal/h /h imp us		km/h	kt	h.min	km	Nm
	75	2500	38	8.4	10.2	237	128	4.55	1178	636
0	65	2350	33	7.3	8.8	220	119	5.40	1248	674
	75	2550	38	8.4	10.2	243	131	4.55	1208	652
2500	65	2400	33	7.3	8.8	225	121	5.40	1288	696
4500	75	2600	38	8.4	10.2	248	134	4.55	1233	666
4500	65	2450	33	7.3	8.8	230	124	5.40	1317	711
0500	75	2650	38	8.4	10.2	254	137	4.55	1263	682
6500	65	2500	33	7.3	8.8	235	127	5.40	1345	727
9500	75	2700	38	8.4	10.2	257	139	4.55	1278	690
8500	65	2550	33	7.3	8.8	240	130	5.40	1375	742
10500	65	2580	33	7.3	8.8	245	132	5.40	1402	757

LANDING PERFORMANCE

At gross weight 1045 kg (2304 lb), Without wind, flaps in "landing" position, engine idling, Dried and plane concrete runway,

PRESSURE	TEMPERATURE °C (°F)	WEIGHT 1045 kg (2304 lb)				WEIGHT 845 kg (1863 lb)			
Zp (ft)			nding tance (ft)	Landing ground roll over 15m(50ft) barrier m (ft)		Landing distance m (ft)		Landing ground roll over 15m(50ft) barrier m (ft)	
0	Std = 15 (59) 35 (95)	230 250 270	(755) (820) (886)	500 530 560	(1641) (1739) (1837)	190 200 215	(623) (656) (705)	425 450 475	(1394) (1476) (1558)
4000	- 13 (7) Std = 7 (45) 27 (81)	260 280 300	(853) (919) (984)	550 585 620	(1805) (1919) (2034)	210 230 240	(689) (755) (787)	465 495 520	(1526) (1624) (1706)
8000	- 21 (-6) Std = -1 (30) 19 (66)	295 320 340	(968) (1050) (1116)	610 650 690	(2001) (2133) (2264)	240 260 275	(787) (853) (902)	510 545 575	(1673) (1788) (1887)

Head wind influence:

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For 30 kt multiply by 0,55

Down wind influence:

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Dried grass runway:

Add 15%